

# A Study on the Feasibility of Stop-and-Go Approximation in FMCW SAR

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## Abstract

Synthetic aperture radar (SAR) specializes in capturing two-dimensional images of Earth's surface. Because satellites or aircraft have mainly been used as SAR platforms, pulse radar systems with high peak transmitted power have been preferred for long-range detection. However, because systems based on pulse radar are generally too heavy and expensive, lightweight and low-cost frequency-modulated continuous-wave (FMCW) radar systems have attracted increasing interest, and many studies on FMCW SAR signal processing are being conducted. The pulse duration of FMCW radar is considerably longer than that of pulse radar. Therefore, it is necessary to determine whether stop-and-go approximation (SAG) is still valid for FMCW radar. If SAG is not applicable, an additional, time-consuming range cell migration correction process is required. In this study, the conditions under which SAG can be applied to FMCW SAR were analyzed. Moreover, Ku-band FMCW SAR field tests were conducted to experimentally validate the feasibility of SAG. Several quantitative parameter values demonstrating the advantages of applying SAG were identified.

**Key Words:** Frequency-Modulated Continuous-Wave (FMCW), Range Cell Migration Correction (RCMC), Stop-and-Go Approximation (SAG), Synthetic Aperture Radar (SAR).

## I. INTRODUCTION

Early synthetic aperture radar (SAR) systems were primarily mounted on satellites. Thus, to cover long ranges between satellites and Earth's surface, pulse radar was preferred. However, due to the increasingly small-scale observation areas and the emergence of SAR platforms such as automobiles and unmanned aerial vehicles, the demand for compact frequency-modulated continuous-wave (FMCW) radar with low-transmission power and cost-effective properties has increased. Accordingly, research on FMCW SAR signal processing algorithms has also increased [1–12].

SAR signal processing aims to determine the exact range and azimuth position of targets by compressing raw data in the range and azimuth direction. Azimuth resolution of SAR image greatly depends on the Doppler (azimuth) bandwidth of the signal. However, range cell migration (RCM) inevitably occurs during SAR data acquisition, resulting in azimuth resolution degradation due to the loss of the Doppler bandwidth. Therefore, before performing azimuth processing, it is necessary to align the azimuth phase information of a target to a single azimuth line to make the most of the Doppler bandwidth for the target. This correction process is called range cell migration correction (RCMC). Fig. 1 shows the bending of the energy trajectory of a single target

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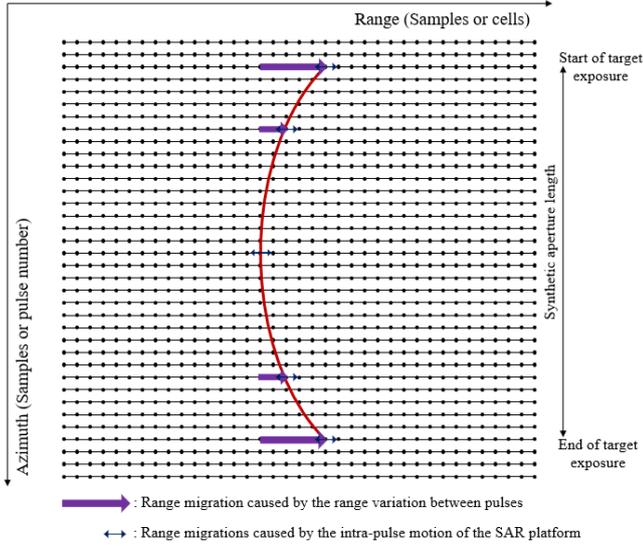


Fig. 1. Visual illustration of a locus (red curve) of energy for a single target according to RCM.

according to RCM.

RCM mainly occurs because the range between the radar and the target varies for each pulse during the SAR integration time. This is the range migration that occurs between pulses. Moreover, since the platform is in motion, frequency shifts due to the relative velocity, and the range variation over the pulse duration may cause additional range migrations. This is the range migration caused by the intra-pulse motion of the SAR platform.

In pulse SAR, the pulse duration is short enough to be assumed that the radar platform is fixed during transmission and reception. Therefore, RCM originating from the latter is negligible in pulse SAR. This assumption is called the stop-and-go approximation (SAG). However, the pulse duration of FMCW radar is relatively long compared to pulse radar. Therefore, it is uncertain whether SAG is still applicable. If SAG cannot be applied, an additional RCMC process is required to compensate for the additional RCM caused by the intra-pulse motion of the SAR platform.

In this study, the range-Doppler algorithm (RDA) was used as a signal processing scheme because it is accurate, efficient, and the most widely used in SAR processing [13]. Fig. 2 shows block diagrams of RDA for FMCW SAR according to whether SAG is applied or not. As shown in Fig. 2(b), the additional RCMC process is omitted when SAG is applied. In general, RCMC involves an interpolation process that requires considerable computational loads [13]. Therefore, the processing time can be greatly reduced if SAG is applied to FMCW SAR.

In some studies, a platform motion during transmission and reception has been considered negligible in the FMCW SAR [2–4]. In other studies, SAG has been considered inapplicable to FMCW SAR [5–8]. Few of these studies have analytically determined whether SAG is applicable, and none have done so

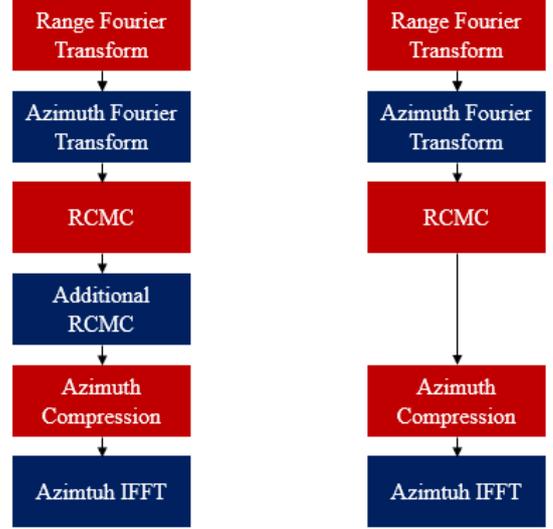


Fig. 2. Block diagrams of RDA for FMCW SAR: (a) RDA without SAG and (b) SAG-applied RDA.

experimentally based on actual raw data [2–8]. Therefore, in this study, the conditions under which SAG can be applied to FMCW SAR were analyzed and experimentally confirmed through Ku-band FMCW SAR field tests.

This paper is organized as follows. In Section II, a review of the FMCW radar signal model is introduced, and the conditions for applying SAG in FMCW SAR are analytically derived. Section III presents the experimental results of Ku-band FMCW SAR field tests. It also presents comparisons of SAR quality parameters and processing times to confirm the applicability of SAG to FMCW SAR and its advantages. Finally, the conclusion of this paper is given in Section IV.

## II. ANALYSIS OF THE VALIDITY OF STOP-AND-GO APPROXIMATION FOR FMCW SAR

This section provides a review of the FMCW radar signal model to determine the range resolution. Linear frequency modulation and sawtooth sweep type are assumed. The transmission signal is

$$s_t(t) = \text{rect}\left(\frac{t}{T}\right) \exp[j2\pi(f_c t + \frac{1}{2}Kt^2)], \quad (1)$$

where  $f_c$  is the carrier frequency,  $T$  is the pulse repetition interval (PRI),  $t$  is the time variable within  $T$ , and  $K$  is the linear chirp rate.  $K$  is given as

$$K = \frac{B}{T}, \quad (2)$$

where  $B$  is the sweep bandwidth of the transmitted signal. The reflected signal from the object with the range  $R$  is determined as

$$s_r(t) = s_t(t - \tau) = s_t\left(t - \frac{2R}{c}\right), \quad (3)$$

where  $\tau$  is the signal round-trip time, and  $c$  is the speed of light. This reflected signal is mixed with a replica of the transmitted signal in the radar receiver. The beat signal, is given as

$$s_b(t) = \exp \left[ j2\pi \left( f_c \tau + K\tau t - \frac{1}{2} K\tau^2 \right) \right]. \quad (4)$$

Since the only phase term related to time  $t$  is the second term, the beat frequency after applying a Fourier transform is calculated as

$$f_{beat} = K\tau = K \frac{2R}{c}. \quad (5)$$

It can be seen that the beat frequency in the FMCW radar receiver is proportional to its distance from the object. The relation between the frequency resolution and T is

$$\Delta f_{beat} = \frac{1}{T}. \quad (6)$$

From Eqs. (5) and (6), the range resolution of the FMCW radar is

$$\Delta R = \frac{c\Delta f_{beat}}{2K} = \frac{c}{2KT} = \frac{c}{2B}. \quad (7)$$

As mentioned in Section I, the pulse duration of FMCW radar is relatively long compared to pulse radar. Therefore, it should be determined whether SAG can be applied to FMCW SAR. Since the platform is in motion during the pulse duration, additional range migrations occur for two reasons. One reason is the frequency shift due to the relative velocity, and the other is the range variation over the pulse duration. For SAG to be applicable to FMCW SAR, these range migrations must be less than the range resolution of the radar.

First, the range migration, caused by the frequency shift due to the relative velocity, is discussed. A radar data acquisition geometry is shown in Fig. 3. For convenience, a zero squint case is assumed.  $L_s$  is the synthetic aperture length,  $\theta_{bw}$  is the antenna beamwidth in the azimuth direction,  $P_0$  is the position of closest approach,  $R_0$  is the range of closest approach, and  $\Delta_s$  is the sample spacing of the synthetic aperture signal. The sample target is continuously detected by the radar while it is in the beam illuminated area.  $P_1$  is the position of the radar when the sample starts entering the beam illuminated area, and  $P_2$  is the position of the radar when the sample exits the beam illuminated area. That is, the sample is detected while the radar is between  $P_1$  and  $P_2$ . Since the distance between  $P_1$  and  $P_2$  is  $L_s$ , the positions of  $P_1$  and  $P_2$  can be specified as points separated by half the  $L_s$  from  $P_0$ . As shown in Fig. 3,  $L_s$  is determined from  $\theta_{bw}$  and  $R_0$  as

$$L_s = 2R_0 \tan \frac{\theta_{bw}}{2}. \quad (8)$$

As the platform moves along the sensor path, the sample is detected by each pulse with a different Doppler frequency. The Doppler frequency shift caused by the relative velocity of the radar and the target is [13]:

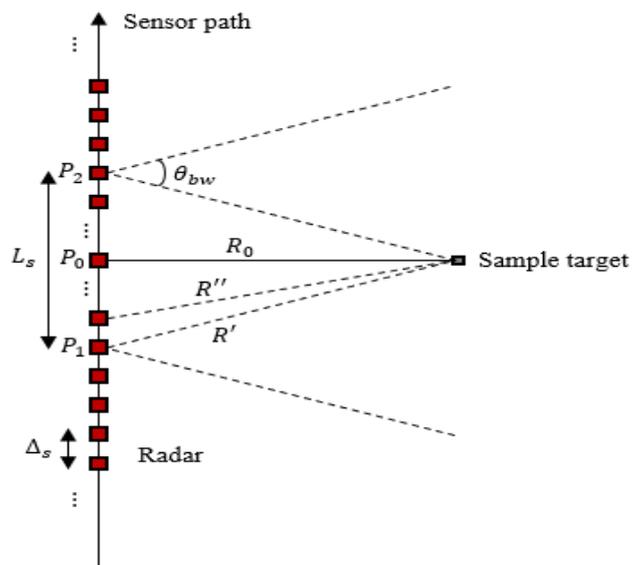


Fig. 3. Radar data acquisition geometry.

$$f_d = \frac{2V_s \sin \theta}{\lambda}, \quad (9)$$

where  $f_d$  is the Doppler frequency shift,  $V_s$  is the radar platform velocity,  $\theta$  is the angle measured from boresight in the slant range plane, and  $\lambda$  is the radar wavelength. As represented in Eq. (5), in FMCW radar processing, the beat frequency is proportional to the range. Therefore, the Doppler frequency shift of the FMCW signal indicates range migration. By substituting Eq. (9) to Eq. (5), the range migration caused by the frequency shift due to the relative velocity can be determined as

$$R_d = \frac{c}{2K} f_d = \frac{fV_s \sin \theta}{K}, \quad (10)$$

where  $R_d$  is the range migration caused by the Doppler frequency shift and  $f$  is the radar frequency. Since the Doppler frequency shift is the largest when the platform is located at  $P_1$  or  $P_2$ , the maximum value of  $R_d$  is determined as

$$R_{d,max} = \frac{fV_s \sin(\theta_{bw}/2)}{K}. \quad (11)$$

Second, the range migration, caused by the range variation over the pulse duration, is discussed. Range migration due to range variation  $R_r$  can be defined as the difference between the range at the pulse start time and the range at the pulse end time (after one PRI). As illustrated in Fig. 3, the range between the radar and the sample when the radar is located at  $P_1$  is  $R'$  which is determined as

$$R' = \sqrt{R_0^2 + \left( V_s \frac{T_s}{2} \right)^2}. \quad (12)$$

$T_s$  is the SAR observation time. Since the platform is in motion, this range changes continuously over the pulse duration. After a PRI, the range between the radar and the sample becomes  $R''$  which is determined as

$$R'' = \sqrt{R_0^2 + \left\{V_s \left(\frac{T_s}{2} - T\right)\right\}^2}. \quad (13)$$

Therefore, the range migration caused by the range variation during the pulse duration can be calculated as

$$R_{r,max} = R' - R''. \quad (14)$$

This is the maximum value of  $R_r$  because when the radar moves to a position other than  $P_1$ , the range migration of the sample is less than  $R_{r,max}$ . Meanwhile, given that  $T_s$  consists of many PRIs,  $R_{r,max}$  is rewritten as

$$R_{r,max} = \sqrt{R_0^2 + \left(V_s \frac{N \cdot T}{2}\right)^2} - \sqrt{R_0^2 + \left\{V_s \frac{(N-2) \cdot T}{2}\right\}^2}, \quad (15)$$

where  $N$  is the number of pulses that constitutes  $T_s$ . When the radar moves from  $P_1$  to  $P_0$  (approaching the sample),  $R_d$  has a positive value, meaning that range migration occurs in the away direction.  $R_r$  also has a positive value, but it means that the range migration occurs in the opposite direction. Therefore, the total range migration is

$$R_m = |R_d - R_r|. \quad (16)$$

On the other hand, when the radar moves from  $P_0$  to  $P_2$  (away from the sample),  $R_d$  has a negative value, meaning that range migration occurs in the forward direction.  $R_r$  also has a negative value, but it means that range migration occurs in the away direction. Therefore, the total range migration is the same as in Eq. (16).

Thus far, the range resolution and the range migration due to the motion of the platform during the pulse have been defined. For SAG to be applicable to FMCW SAR, the total range migration  $R_m$  should be less than the range resolution of the radar  $\Delta R$ . Thus, the following condition must be satisfied:

$$R_m < \Delta R. \quad (17)$$

Fig. 4 shows the range migration (solid lines) as a function of frequency at different radar platform velocities and the range

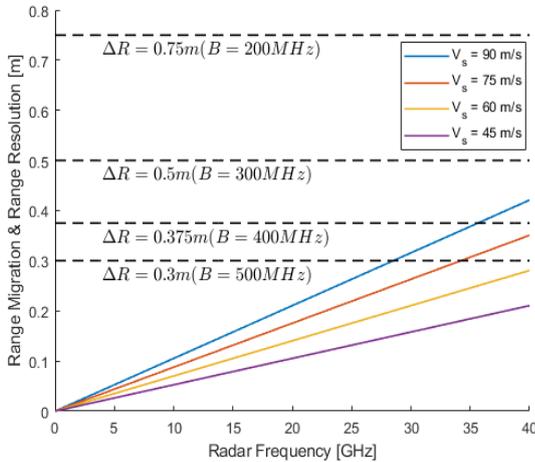


Fig. 4. Range migration and range resolution according to FMCW SAR system specifications.

resolution (dashed lines) as a function of frequency bandwidth. For simplicity, a linear chirp rate of  $K = 2.5$  THz/s and an antenna beamwidth of  $\theta_{bw} = 34^\circ$  are assumed. In addition, as will be discussed in Section III,  $R_r$  is not considered because its value is almost zero. On the other hand, for a fixed value of  $K$ , as  $T$  decreases,  $B$  also decreases, and  $\Delta R$  degrades (i.e., widens). This is in line with the fact that SAG is valid for pulse radar with a short pulse duration. Notably, a value of  $V_s = 90$  m/s is almost the fastest velocity available in FMCW SAR [1, 4, 7, 9–12]. This is because FMCW SAR is used only on aircraft or automobile platforms due to its detection range limitation.

As shown in Fig. 4, range migration increases as the range frequency and radar platform velocity increases. Also, the wider the frequency bandwidth, the better (narrower) the range resolution. Therefore, the most difficult case to apply SAG to FMCW SAR is when the system has high-frequency, high platform velocity, and high range resolution properties. By comparing the expected range migration with the range resolution, it is possible to determine whether SAG is applicable to FMCW radar with certain system specifications. SAG can be applied to FMCW SAR in the Ku-band frequency region (12–18 GHz), even in the case of high  $\Delta R$  (0.3 m) and high  $V_s$  (90 m/s) because the range migration in that region is considerably smaller than the range resolution. On the other hand, SAG is not applicable to FMCW SAR in the Ka-band frequency region (27–40 GHz) in the case of high  $\Delta R$  (0.3 m) and high  $V_s$  (90 m/s) and can be applied only if  $V_s$  decreases or  $\Delta R$  widens. Therefore, the system specifications should be considered to determine whether SAG is applicable to a given FMCW SAR system.

Section III presents range migration and range resolution calculations with actual Ku-band FMCW SAR system specifications. Also, it presents field test demonstrating that SAG is applicable to Ku-band FMCW SAR even in tough cases.

### III. EXPERIMENTAL VALIDATION OF THE FEASIBILITY OF STOP-AND-GO APPROXIMATION IN KU-BAND FMCW SAR THROUGH FIELD TESTS

The detailed specifications of the Ku-band FMCW SAR system used in the outdoor experiments are given in Table 1. The center frequency of the FMCW radar is in the Ku-band (14.25 GHz). The frequency bandwidth of the radar system is 500 MHz, which leads to a high  $\Delta R$  (0.3 m). To avoid the limitations of a single scenario, experiments were performed at various radar platform velocities (82, 75, 63, and 45 m/s). In Fig. 5, since the expected range migration values are below the range resolution line, SAG is applicable to all cases. The applicability of SAG to a specific FMCW SAR system can be confirmed by directly calculating the additional range migration and the range resolution using

Table 1. Specifications of the Ku-band FMCW SAR system

System parameters	Value
Center frequency	Ku-band (14.25 GHz)
Frequency bandwidth	500 MHz
Linear chirp rate	2.5 THz/s
PRI	200 $\mu$ s
Linear FM sweep type	Sawtooth
Transmission power	39 dBm
Antenna gain	16 dBi
Antenna beamwidth	34 $^\circ$

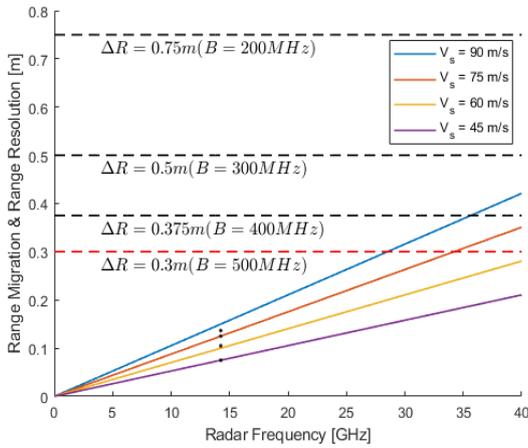


Fig. 5. Expected range migrations (black dots) and range resolution (dashed red line) for given specifications.

the equations presented in Section II. The case of a radar platform velocity of 75 m/s is considered below.

In our system specification, using Eq. (11), the maximum range migration caused by the frequency shift due to the relative velocity  $R_{d,max}$  was 0.125 m, which was less than the range resolution  $\Delta R$ . This value was consistent with the expected range migration value for the case of  $V_s = 75$  m/s in Fig. 5. Moreover, since  $T_s$  consists of thousands of PRIs,  $R_r$  is almost zero and, therefore, negligible. In fact, using Eq. (15), the maximum range migration  $R_{r,max}$  caused by the range variation over the pulse duration was about 0.0044 m, which is considerably less than the range resolution. This means that SAG was applicable to our FMCW SAR system case, as shown by the calculation of Eq. (17).

To experimentally verify the applicability of SAG to FMCW SAR, SAR signal data were obtained from several test sites at various platform velocities. Appropriate SAR images appeared after processing SAG-applied RDA in the actual raw dataset. Fig. 6 shows the Ku-band FMCW SAR images. The SAR images match the aerial photographs of the test sites shown in Fig. 7, providing experimental evidence that our specific Ku-band FMCW SAR can work with SAG.

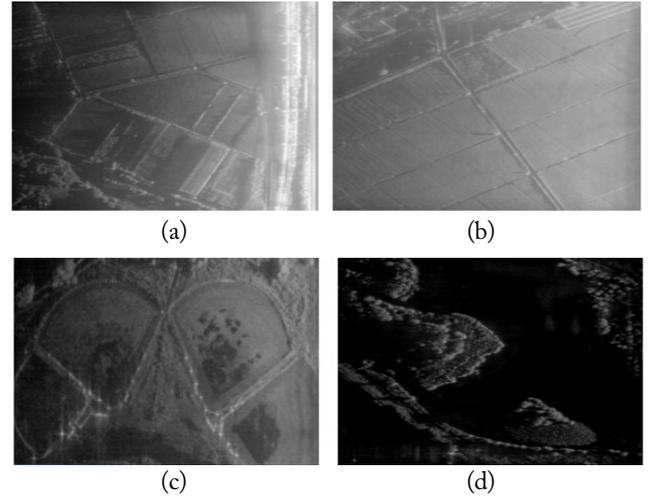


Fig. 6. Ku-band FMCW SAR images: (a) 75 m/s, (b) 75 m/s, (c) 45 m/s, and (d) 82 m/s.

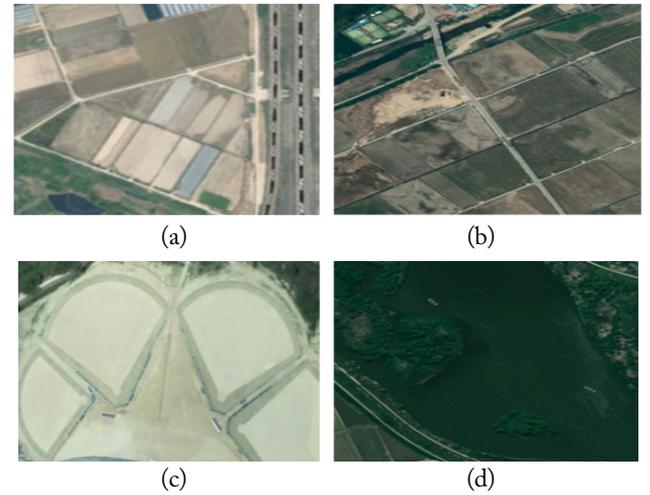


Fig. 7. Aerial photographs of Ku-band FMCW SAR test areas: (a) 75 m/s, (b) 75 m/s, (c) 45 m/s, and (d) 82 m/s.

Moreover, an impulse response function (IRF) analysis was performed to provide a quantitative basis for the validity of SAG for Ku-band FMCW SAR. Essential SAR quality parameters, such as impulse response width (IRW) and peak sidelobe ratio (PSLR), were estimated from the IRF [13]. Fig. 8(a) shows a trihedral corner reflector (CR) installed at the field test site for the IRF analysis. Fig. 8(b) shows an aerial photograph of the CR test site. To determine whether SAG degraded SAR image quality, RDA without SAG (Fig. 2(a)) and with SAG (Fig. 2(b)) were performed and compared. Fig. 9(a) and 9(b) shows the results of the FMCW SAR images for each case. The CR clearly emerged inside the red circles shown in the images. The IRF of the CR in Fig. 9(a) is shown in Fig. 10(a)–(c). For visual clarity, an interpolation by a factor of 16 was implemented. Fig. 10(a) shows a normalized intensity of the IRF in 3D, and Fig. 10(b) and 10(c) shows one-dimensional profiles of the IRF in the range

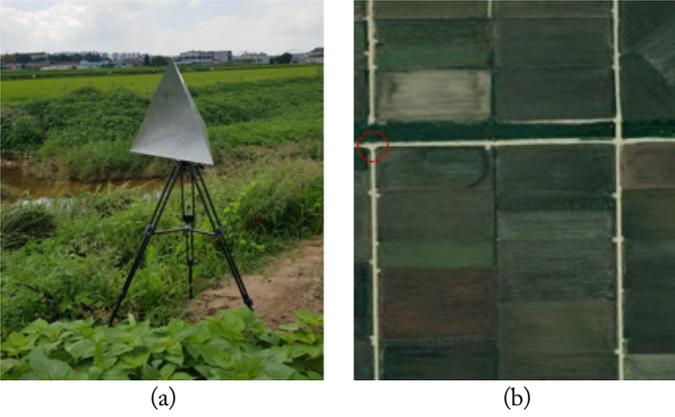


Fig. 8. (a) A trihedral CR and (b) an aerial photograph of the field test site. The CR is installed inside the red circle.

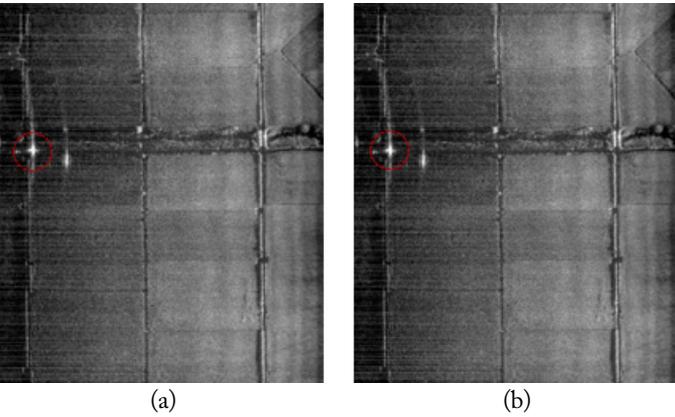


Fig. 9. Ku-band FMCW SAR images ( $V_s = 63$  m/s): (a) RDA without SAG and (b) SAG-applied RDA.

and azimuth direction, respectively. Likewise, the IRF of the CR in Fig. 9(b) is shown in Fig. 10(d)–(f). The peak of the signal was well represented in both the range and azimuth directions. The SAR quality parameters and processing times of RDA with and without SAG are summarized in Table 2. As shown in Fig. 10 and Table 2, applying SAG did not affect the IRW or PSLR values. This provides clear quantitative evidence that our specific Ku-band FMCW SAR can work with SAG. Furthermore, the processing time was significantly reduced (by about 36%) when SAG was applied. This suggests that SAG shortened the processing time without SAR quality parameter performance loss.

#### IV. CONCLUSION

A major contribution of this study is its analytical and experimental approaches to determining the feasibility of SAG for FMCW SAR signal processing, which has been contested. No

Table 2. Comparison of SAR quality parameters and processing times

SAR quality parameter	IRW (sample)		PSLR (dB)		Processing time (s)
	Range	Azimuth	Range	Azimuth	
RDA without SAG	2.13	2.94	-5.14	-6.41	114.16
SAG-applied RDA	2.13	2.94	-5.11	-6.39	72.01

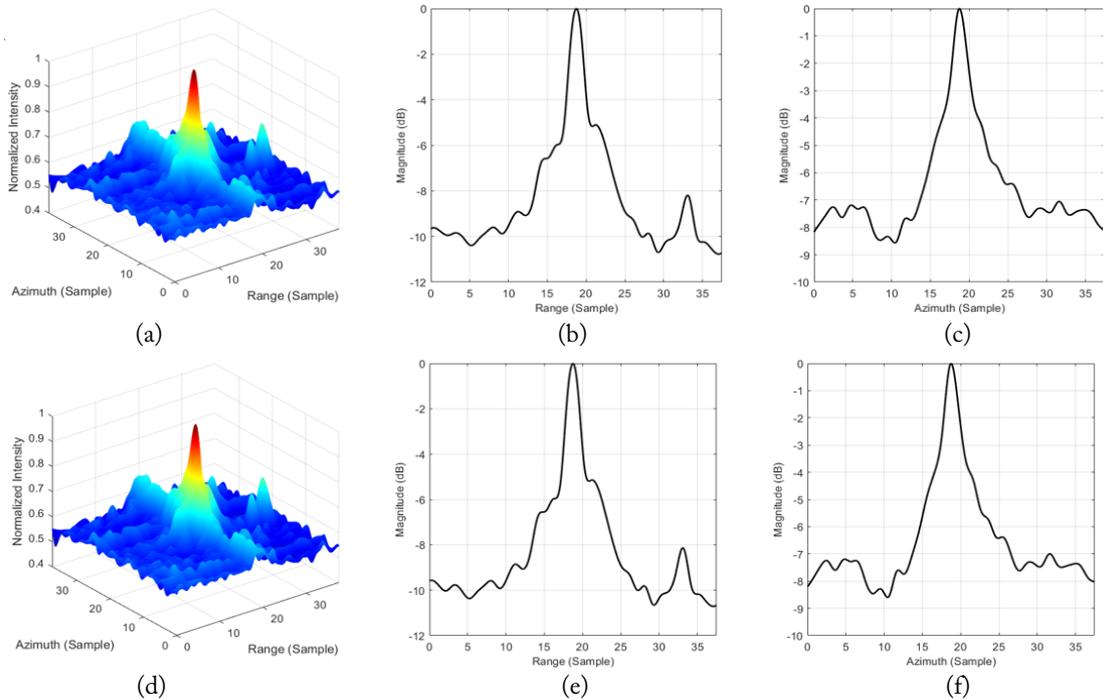


Fig. 10. IRF of the CR: (a) 3D IRF (RDA without SAG), (b) range profile (RDA without SAG), (c) azimuth profile (RDA without SAG), (d) 3D IRF (SAG-applied RDA), (e) range profile (SAG-applied RDA), and (f) azimuth profile (SAG-applied RDA).

previous studies have verified the validity of SAG using practical raw data. Therefore, in this study, the conditions for applying SAG to FMCW SAR were analyzed and experimentally verified for Ku-band FMCW SAR. The results show that if SAG is applicable, the processing time is reduced without compromising the SAR image quality. Since the platform velocity used was at the aircraft level, the results may be generalizable to automobile applications as well. Moreover, since the operating frequency used in the experiment was in the Ku-band (14.25 GHz), SAG can also be applied to FMCW SAR using operating frequencies below the Ku-band if other parameters are similar. Future studies could investigate whether SAG is applicable to FMCW SAR using frequencies above the Ku-band, such as the K- or the Ka-band.

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